King Fahd University of Petroleum & Minerals MECHANICAL ENGINEERING DEPARTMENT

ME 422 : PROPULSION SYSTEMS

AE 422 : FLIGHT PROPULSION I

Spring Semester 2000-2001 (002)

Instructors:	Dr. B. S. Yilbas	Office: 22-230	Phone: 4481
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Textbook:MECHANICS AND THERMODYNAMICS OF PROPULSION
SYSTEMS. by Hill, P. and Peterson, C., Addison-Wesley Pub. Ltd., 1992.

References: 1. An Introduction to Aerospace Propulsion by Archer, R. D. and Saarlas, M., Prentice Hall Publisher, 1996.
2. Aerothermodynamics of Gas Turbine and Rocket Propulsion by Oates, G. C.

Course Description:

Aerothermodynamics of aerospace vehicle engines, combustion, thrust and efficiency. Gas turbine engines: Turbojet, turbofan, turboprop ; Ramjet and scramjet , typical engine performance. Aerothermodynamics of inlets, combustors and nozzles. Introduction to propellers, turbocompressors and turbines. Introduction to rockets and performances of rocket vehicle engines. Chemical and electrical driven rocket engines.

Prerequisites: ME 204, ME 311

Wee k	Classes (50	Chapters	Topics	
ĸ	(30 min)			
1	3	2	Mechanics and Thermodynamics of Fluids.	
2,3	6	5	Thermodynamics of Aircraft Jet Engines.	
4	3	6	Aerothermodynamics of Inlets, Combustors and Nozzles.	
5,6,7	9	7	Axial Compressors.	
8,9	6	8	Axial Turbines.	
10,11	6	9	Centrifugal Compressors.	
12,13	6	10	Performance of Rocket Vehicles.	
14,15	6	11,12	Chemical Rocket Thrust Chambers and Propellants.	

Material to be covered:

Evaluation:

Quizzes	18%	
Major Exam 1	17%	24 th March 2001 (6:00-8:00pm)
Major Exam 2	22%	5 th May 2001 (6:30-8:30pm)
Assignments	8%	
Final Exam	35%	

Attendance:

University regulations on attendance will be strictly enforced.

Homework: Each student must submit his homework solution on time (no late homeworks will be accepted). All homeworks solutions should have the Department standard cover sheet in the front .

Homework	1	2	3	4	5	6	7
Problem #	5.2,5.7,5.9,	6.1,6.2,6.7,	7.2,7.3,7.6,	8.2,8.5,8.8,	9.2,9.5,9.7,	10.2,	11.1,11.3,
(from textbook)	5.15,5.17,	6.11,6.15	7.11,7.13	8.10,8.13,	9.12,9.15	10.4	11.8,12.3,
•	5.20			8.17			12.7